Errata

Motion and Deformation of Very Large Space Structures

Ramesh B. Malla
University of Connecticut, Storrs, Connecticut
and

William A. Nash and Thomas J. Lardner University of Massachusetts, Amherst, Massachusetts

[AIAAJ 27, No. 3, pp. 374-376 (1989)]

THE following errors were made inadvertently during production of this paper. On page 375, Eqs. (3a) and (3c) contained errors. The correct equations are as follows:

$$m\ddot{r}_{c} - mr_{c}\dot{\theta}^{2} + \mu \left[\frac{m_{1}(r_{c} + x_{1}\cos\phi)}{r_{1}^{3}} + \frac{m_{2}(r_{c} - x_{2}\cos\phi)}{r_{2}^{3}} \right] = Q_{r}$$
(3a)

$$\bar{m}\ddot{x} - \bar{m}x(\dot{\theta} + \dot{\phi})^2 + \bar{m}\mu \left[\frac{(x_1 + r_c \cos\phi)}{r_1^3} + \frac{m_2(x_2 - r_c \cos\phi)}{r_2^3} \right]$$

$$+\frac{Gm_1m_2}{x^2} + \frac{\partial U_e}{\partial x} = Q_x \tag{3c}$$

On page 376, the penultimate sentence in the section titled "Physical Model and Initial Conditions" should read as follows:

Two elliptical orbits are considered—one with the initial orbital angular velocity $\dot{\theta}_0$ equal to 0.0738 rad/min, giving a small orbit eccentricity (e = 0.0785), and the other with $\dot{\theta}_0$ equal to 0.08883 rad/min, giving a fairly large orbit eccentricity (e = 0.56).

Control of Flexible Structures by Applied Thermal Gradients

Donald L. Edberg

Jet Propulsion Laboratory,

California Institute of Technology,

Pasadena, California

[AIAAJ 25, No. 6, pp. 877-883 (1987)]

THE following errors were made inadvertently during production of this paper. On page 878, Eqs. (2) and (5) contained errors. The correct equations are as follows:

$$T = \frac{\partial T_{\text{step}}(t)}{\partial t} = \frac{Q}{\rho c h S} \left[1 + \frac{2}{\pi^2} \sum_{n=1}^{\infty} \frac{(-1)^n}{n^2} \beta_n e^{-\beta_n t} \cos \frac{n \pi z}{h} \right]$$
(2)

$$\frac{T(t)}{V_c(t)} = \frac{C_{QV}}{\rho c h n_s S} \left[1 + \frac{2}{\pi^2} \sum_{n=1}^{\infty} \frac{(-1)^n}{n^2} \beta_n e^{-\beta_n t} \cos \frac{n \pi \mathcal{Z}}{h} \right]$$
 (5)